

Patent Application of

5

**Kenneth A. Honer**

For

**Sputtered Silicon for Microstructures and Microcavities**

10

*pro. →  
h*

**STATEMENT REGARDING FEDERALLY SPONSORED RESEARCH OR  
DEVELOPMENT**

15

This invention was supported in part by contract number F30602-96-2-0308-P0001 from the Defense Advanced Research Projects Agency (DARPA) Composite CAD program and the Air Force Office of Science and Research (AFOSR). The U.S. Government has certain rights in the invention.

20

**FIELD OF INVENTION**

25

This invention relates generally to fabrication of silicon micro mechanical structures. More particularly, it relates to released microstructures made of sputtered silicon that are compatible with pre-fabricated on-chip aluminum circuitry.

30

000111 68401260

## BACKGROUND OF INVENTION

Polysilicon has become the material of choice for most surface micro-machined structures due to its excellent mechanical properties and controllable stress. The commonly used process to deposit polysilicon is low pressure chemical vapor deposition (LPCVD), which requires relatively high deposition temperatures of 580-630°C and annealing temperatures above 900°C. See for instance R.T. Hove and R. S. Muller, Abstract No. 118, Extended Abstracts of the Electrochemical Society Meeting, Montreal, Canada, May, 1982, pp. 184-185; or R. T. Howe and R. S. Muller, "Integrated Resonant-Microbridge Vapor Sensor" Proc. IEEE Int. Elect. Dev. Mtg., San Francisco, Dec., 1984, pp. 213-216.

Since electronic circuitry is typically heat resistant only up to temperatures below those required for the LPCVD, re-engineering may have to be performed to bring pre-fabricated circuitry into operational condition unless the metal layers are deposition after the polysilicon deposition as is described for example in J. H. Smith, et al., "Embedded micromechanical devices for the monolithic integration of MEMS with CMOES," Proc. Int. Electron Devices Mtg., Washington, DC, Dec., 1995, pp. 609-612.

Another alternative is to use refractory metals such as tungsten instead of aluminum as is described, for example in J. M. Bustillo, et al., "Process technology for modular integration of CMOS and polysilicon microstructures," Microsystem Technologies, Vol. 1, 1994, pp.30-41. These approaches increase the overall fabrication complexity.

The integration of microstructures and electronic circuitry is vital to the performance of many surface micro-machined sensors, since as dimensions decrease, sensitivity often  
5 falls off precipitously. The sensitivity of a torsional capacitive accelerometer, for example, scales with the fifth power of the lateral dimension of it.

10 Stresses and strain gradients can limit the performance of both integrated and passive electrostatic MEMS devices. If the in-plane residual stress in doubly supported structures, for example, is too large, the structures may buckle. Conversely, if the stresses are too large the mechanical stiffness may be too large for the intended  
15 application. Stress control in sputtered structures has been achieved previously with high temperature anneals as is described for example, in T. Abe and M. L. Reed, "Low Strain Sputtered Polysilicon for Micromechanical Structures," Proc. of Ninth International Workshop on Micro  
20 Electro-Mechanical Systems, San Diego Feb., 1996. These high temperature anneals (>1000C) exceed their thermal annealing budget critical thermal budgets of integrated circuitry. The critical thermal budget is the budget beyond which the configuration of the integrated circuitry  
25 becomes permanently changed.

The presence of strain gradients in deposited films causes released structures to warp. If the warpage is severe, the structures may touch the substrate, rendering them  
30 immobile. Alternatively, if the structures warp away from the substrate, capacitances of the structures diminish. Strain gradients result from initially sputtered silicon

09710489-111000

sub.  
163

forming clusters on top of the sacrificial layer before reaching a closing thickness at which the sputtered silicon clusters have laterally extended and sufficiently overlapped to form a solid layer. The closing thickness is within  
5 several 100 Angstroms. The sputtered silicon within that closing thickness has a specific internal stress that differs significantly from the internal stress of the silicon deposited above the closing thickness. The phenomenon of the differing internal stress within the  
10 closing thickness is known to those skilled in the art closing of the initial clusters<> as coalescence. The closing thickness is highly constant and introduces an essentially constant coalescence strain to the final structure. With increasing overall thickness of the  
15 micromachined structure the influence of the coalescence strain becomes less influential. Depending upon the application, several additional properties may be important for the structural layer of a micro-fabricated device. Among these characteristics are  
20 film density, surface roughness and electrical resistivity, and permeability.

Thin layers of polysilicon are permeable to HF based etches as is described, for example, in K. S. Leboutitz, R. T.  
25 Howe, and A. P. Pisano, "Permeable Polysilicon Etch-Access Windows for Microshell Fabrication" Transducers '95, pp. 224-227. Unfortunately, the layer thickness useful for creating solid enclosed cavities are too thin for extended mechanical application and require additional reinforcing  
30 layers. In particular, the capillary force of the etch requires a minimal thickness of the permeable layer to prevent a collapsing of the cavity cover. To prevent

collapsing during wet etching, a critical point drying has to be performed, where the wet etch is frozen and then evaporated.

- 5 The inventive utilization of low temperature sputtering techniques for depositing silicon layers makes the use of organic materials for sacrificial layers possible. Such an organic material is preferably polyimide, which can be etched and removed by the use of a dry etch or dry plasma  
10 etch where capillary forces do not occur.

#### OBJECTS AND ADVANTAGES

- 15 It is a primary object of the present invention to provide a micro-machined structure that can be fabricated non destructively to and in combination with pre-fabricated aluminum-metalized electronic circuitry.

- 20 It is another object of the present invention, to provide a method for making the micro-machined structure with tunable in-plane strain and strain gradient.

- It is a further object of the present invention, to provide  
25 a micro-machined structure with predetermined levels of electric conductivity.

- It is also an object of the present invention to provide a micro-machining process that employs a dry-release of the  
30 resultant micro-structure.

Finally, it is an object of the present invention, to provide a micro-machined structure with a porosity for creating encapsulated cavities with sufficiently thick cavity cover layers.

5

### SUMMARY

10 A sputtered layer is introduced. The sputtered layer, preferably from silicon, is deposited with predetermined sputtering criteria resulting in a predetermined resulting<> or pre-annealing configuration. This pre-annealing configuration is transformed during a low temperature annealing process into a post-annealing configuration. A released structure is micro-machined from the sputtered layer in its post-annealed configuration.

20 The fabrication process includes the initial deposition of a sacrificial layer, for example, phosphosilicate glass (PSG) or polyimide resin beneath the following layers predetermined for the released structure. The sacrificial layer is removed in a final fabrication step to the extent necessary to release the previously patterned structure.

25 The released structure has a resulting in-plane strain and a resulting strain gradient, which are predetermined in accordance with deformation configurations of the released structure. The deformation configuration depends on the shape of the released structure and on the fashion it is supported. The deformation configurations are distinguished between an essentially buckling-free deformation

configuration and an essentially buckling-influenced deformation configuration. The buckling-free deformation configuration is, for example, a beam supported on one end. The buckling influenced deformation configuration is, for example a straight beam rigidly supported on both ends. In other words, the buckling-free deformation configuration is the case where the buckling principles according to Euler are not applicable; the buckling-influenced deformation configuration is the case where the buckling principles according to Euler have to be applied.

In the case, where a low temperature annealing is included in the fabrication of the released structure, the resulting in plane strain and the resulting strain gradient are only an initial in-plane strain and an initial strain gradient, which are transformed during the annealing into a residual in-plane and a residual strain gradient. Resulting in-plane strain / strain gradient and initial in-plane strain / strain gradient may differ, since the low temperature annealing allows an additional adjustment of them independently from the sputtering. Thus, sputtering criteria defined to deposit the silicon, may be more broadly selected when a low temperature annealing is included in the fabrication of micro-machined structures. The low temperature annealing is an optional fabrication step.

For buckling-free deformation configurations, first sputtering criteria predominantly include sacrificial layer etchant selection and sputtering layer thickness are defined such that the sputtered layer has a predetermined resulting or initial strain gradient. In the case, where a

low temperature annealing is included in the fabrication process, the first sputtering criteria are selected in correlation with the annealing transformation.

5 For buckling-influenced deformation configurations, second sputtering criteria predominantly include and sacrificial layer material are defined such that the sputtered layer has a predetermined resulting or initial in-plane strain. Sputtering power, ambient sputtering pressure and sputtering temperature are selected from a zone-T of the Thornton zone diagram as known to those skilled in the art. In the case, where a low temperature annealing is included in the fabrication process, the second sputtering criteria are selected in correlation with the annealing transformation.

The low temperature annealing is performed with a thermal annealing budget, which includes sufficiently low maximum temperatures and sufficiently short annealing duration such that pre-fabricated integrated electronic circuitry is not permanently altered by the annealing. Thus, integrated electronic circuitry like, for example, aluminum-metalized circuitry or aluminum-metalized CMOS, can be fabricated and brought into operational condition on the same work piece or chip prior to the making of the micro-machined structure.

To adjust the electric conductivity of the micro-machined structure, a multi layer including a silicon core layer and at least one conductive layer with high electric conductivity are combined. The conductive layer has a dissolving characteristic that is compatible with the



dissolving characteristic of the core layer. The dissolving characteristic includes an etching rate at which material of the multi layer structure is removed for a first group of etchants, selected to etch the multi layer structure.

- 5 The dissolving characteristic includes further an etchant resistance against a second group of etchants, selected to etch the sacrificial layer. The compatible dissolving characteristic of the individual layers of the multi layer allows a simultaneous etch forming of the released
- 10 structure from the multi layer. In the case where the core layer is made from silicon, the secondary layer(s) may be a Titanium based material such as TiW or TiN.

000TTT" 6340T260

15 *sub. 1/67* The sputtered layer made from silicon is porous and permeable to HF-based etches at approximately ten times the thicknesses reported for LPCVD deposited polysilicon. The initial porosity remains mostly unaltered during the annealing process such that encapsulated cavities with relatively thick cover layers may be fabricated compared to

20 those made from polysilicon.. Consequently, larger and/or more solid cover layers may be fabricated compared to those from prior art methods. The cover layers are stiffer and better able to withstand the capillary forces of the wet etch in the encapsulated cavity beneath the silicon during

25 a drying process. The annealing need not be performed prior to the release etching.

#### BRIEF DESCRIPTION OF THE FIGURES

30

**Fig. 1** shows a simplified section of a work piece during the sputtering process.

**Fig. 2** shows a simplified section of the work piece of **Fig. 1** during the annealing process.

5 **Fig. 3** shows a simplified section of the work piece of **Fig. 1** after the patterning of micro-machined structures.

**Fig. 4** shows a simplified section of the work piece of **Fig. 1** after the releasing of the micro-machined structures of  
10 **Fig. 3**.

**Fig. 5** shows a simplified block diagram showing the fabrication steps of the present invention.

15 **Fig. 6** shows a simplified section of a work piece with aluminum-metalized circuitry elements with a sacrificial layer and an aluminum hard mask on top of it.

**Fig. 7** shows a simplified section of the work piece of **Fig. 6** after a primary patterning of the vias in the sacrificial layer used for anchoring the microstructure to the  
20 substrate.

**Fig. 8** shows a simplified section of the work piece of **Fig. 7** after the sputtering of structural films on top of a  
25 patterned sacrificial layer. Three films are shown. A primary structural layer and two secondary layers are used to improve conductivity, sculpt stress, in-plane strain and strain gradient.

30

**Fig. 9** shows a simplified section of the work piece of **Fig. 8** after the release of the micro-machined multi-layer structure.

5

#### DETAILED DESCRIPTION

Although the following detailed description contains many specifics for the purposes of illustration, anyone of ordinary skill in the art will appreciate that many variations and alterations to the following details are within the scope of the invention. Accordingly, the following preferred embodiment of the invention is set forth without any loss of generality to, and without imposing limitations upon, the claimed invention.

Fabrication processes for micro-machined structures according to the present invention are illustrated in the **Figs. 1-4** and **6-9**. The contents of **Figs. 1-4** and **6-9** is simplified and without any claim of dimensional or proportional accuracy and serves solely to give an understanding of the inventive steps of the present invention.

The fabrication example described in the **Figs. 1-4** is for a fabrication process including a low temperature annealing. It is clear to one skilled in the art that the scope of the invention is not limited by the inclusion of a low temperature annealing.

30

**Fig. 1** shows a simplified section of a work piece during the sputtering process. The work piece may be a substrate

or any structure including a substrate known to those skilled in the art as a base for micro-machined structures and electronic circuitry. The work piece is a unit undividedly exposed to the fabrication steps required for fabricating micro-machined structures including a thermal annealing process. The work piece may include pre-fabricated electronic circuitry having a critical thermal budget beyond which the circuitry configuration becomes permanently altered. The critical thermal budget is defined by exposure temperature and exposure duration as is known to those skilled in the art.

The simplified section of **Fig. 1** is shown between the breaking lines **B1** and **B2** having a substrate **1** with a sacrificial layer **2** on top of the substrate **1**. On top of the sacrificial layer **2** is the initial sputtered layer **3A** deposited by the sputtering process **Sp**.

The sputtering process **Sp** is either defined by first sputtering criteria **P1** or by second sputtering criteria **P2**. The first sputtering criteria **P1** provide an initial sputtered layer **3A** having a first pre-annealing configuration including a predetermined initial strain gradient. The second set of sputtering criteria provides the initial sputtered layer **3A** having a second pre-annealing configuration including a predetermined initial in-plane strain. The sputtering process should be of zone-T type to ensure that low stresses are achieved. The deposition temperature should thus be approximately between room temperature and 200C. An Ar working gas deposition pressure ranging from 8 to 14 mTorr yielded acceptable stresses. Films deposited at 8 mTorr were more dense. Those

deposited at 14 mTorr were more porous. Deposition powers of between 1.5 and 2.5 kW were used.

Please note that much of these results are machine specific, which may vary between various types of sputtering machines as is well known to those skilled in the art. Zone-T type sputtering is a sputtering according to the Thornton zone diagram as is well known to those skilled in the art.

**Fig. 2** shows a simplified section of the work piece of **Fig. 1** during the annealing process. The simplified section of **Fig. 2** features optional aluminum terminals **Al**, which may be deposited after the deposition of the initial sputtered layer **3A** and before the work piece is exposed to a low temperature annealing process **TAn** indicating in **Fig. 2** a rectangle labeled **TB** for a thermal annealing budget **TB** induced on the work piece during the low temperature annealing process **TAn**. The definition of the thermal annealing budget **TB** includes a maximum annealing temperature and an annealing duration. The thermal annealing budget **TB** is smaller than the critical thermal budget of eventual electronic circuitry of the work piece simultaneously exposed to the low temperature annealing process **TAn**. The films were annealed in nitrogen and a nitrogen-hydrogen environment at temperatures at or below 350C. The films annealed in the nitrogen-hydrogen environment showed a marked decrease in conductivity and a decrease in the strain gradient.

During the low temperature annealing process **TAn**, the initial sputtered layer **3A** is transformed into a layer **3T** having a transformed structure.

**Fig. 3** shows a simplified section of the work piece of **Fig. 1** after the patterning of micro-machined structures **31**. The patterning is accomplished by a first etching through the gaps of an etching mask **5** deposited after the low temperature annealing process **TAn**. The micro-machined structures **31** are shaped from the residual sputtered layer **3Z**, which is the result of the transformed layer **3T** after the low temperature annealing process **TAn**.

The residual sputtered layer **3Z** may have predetermined first post-annealing configuration with a residual strain gradient that results from the first pre-annealing configuration in combination with the predetermined transformation taking place in the transforming layer **3T**.

The residual sputtered layer **3Z** may have a predetermined second post-annealing configuration with a residual in-plane strain that results from the second pre-annealing configuration in combination with the predetermined transformation taking place in the transforming layer **3T**.

The residual sputtered layer **3Z** is patterned using standard semiconductor processing techniques.

**Fig. 4** shows a simplified section of the work piece of **Fig. 1** after releasing the micro-machined structures **31** of **Fig. 3**. The release is accomplished by dissolving the release area **22** of the sacrificial layer **2** in an etching process

that preferentially effects only the sacrificial layer 2. Consequently, the remaining contacting surfaces 32 are not contacting the sacrificial layer 2 anymore and the micro-machined structures 31 are released.

5

The micro-machined structures 31 may have either an essentially buckling-free deformation configuration or an essentially buckling-influenced deformation configuration. In case of the buckling-free deformation configuration, the likelihood for buckling of the released structures 31 is neglect able. In case of the buckling-influenced deformation configuration, the likelihood of buckling according to the buckling laws of Euler is a dominant deformation threat of the released micro-machined structures 31.

10

15

For buckling-free deformation configurations, first sputtering criteria are selected.

20

For buckling-influenced deformation configurations, second sputtering criteria are selected.

25

It is clear to one skilled in the art that the low temperature annealing process may be performed not only before the first etching process but also prior or after other above described fabrication steps in a feasible fashion.

30

In the preferred embodiment of the invention, the substrate 1 is made from silicon and the initial sputtered layer 3A is made from silicon preferably doped with 40-80 ppm boron. The boron doping improves electrical conductivity and

increases the sputtering rate. The substrate 1 may be made from silicon alone. The electronic circuitry may include aluminum-metalized semiconductors such as CMOS.

5 The sacrificial layer 2 may be preferably phosphosilicate glass (PSG) with a phosphorous content of 8%. The sacrificial layer 2 may also be polyimide resin, undoped low-temperature oxide (LTO) or any other material known to those skilled in the art being suitable as sacrificial  
10 layer material. PSG is preferably selected for the sacrificial layer 2 for wet released processes because of its higher HF etch rate relative to undoped low temperature oxide. Since the sputtered silicon is permeable to HF-based etchants, there need be no direct path to the underlying  
15 sacrificial oxide layer for etching to take place.

In the extreme case, where another micro-machined structure forms a solid cover on top of a dedicated released area 22, the etching solvent reaches the sacrificial layer through  
20 the porous structure of the residual sputtered layer 32.

HF-based etchants are preferably 6:1-20:1 buffered HF or  $\text{NH}_4\text{HF}+\text{HC}_2\text{H}_3\text{O}_2+\text{H}_2\text{O}$  (pad etch or PAD). The permeability rate for these HF-based etchants through sputtered silicon is  
25 shown in the first table below. The permeability rate is related to the time needed to etch the underlying sacrificial layer of PSG. The first table below shows, that sacrificial layers are dissolvable through sputtered silicon at least 5.0  $\mu\text{m}$  thick.

30

thickness	of	Time for 6:1	Time for 20:1	Time
-----------	----	--------------	---------------	------



sputtered silicon membrane	buffered HF	buffered HF	for PAD
0.6 $\mu\text{m}$	15 min	55 min	20 min
2.0 $\mu\text{m}$	15 min	55 min	20 min
5.0 $\mu\text{m}$	15 min	55 min	20 min

After creating the released area **22**, which may be an enclosed cavity, the permeability of the residual sputtered layer **3Z** may be compensated by coating the residual  
5 sputtered layer **3Z** with a sealant like, for example,  $\text{Si}_3\text{N}_4$ .

Where sufficient access is given to the sacrificial layer **2**, a dry etch such as oxygen plasma may be used to create the released area **22**. In such a case, polyimide resin may  
10 be used for the sacrificial layer **2**. This has advantages over oxide sacrificial layers in that the CMOS circuitry does not need to be protected during the release and special drying techniques are not required to prevent capillary forces from permanently pulling the  
15 microstructures to the substrate.

The initial sputtered layer **3A** may be deposited with a predetermined sputtered layer thickness, which influences the initial strain gradient. Initial and residual strain  
20 gradient are further influenced by the selected etchant used in the later fabrication step to remove the sacrificial layer **2** in order to provide the released area **22**. The HF etch attacks the grain boundaries, changing the stress at those points. This strain altering effect that  
25 takes place during HF etching is indicated with the surrounding rectangle label with "Hfbased". The following

second table shows strain gradients expressed as curvature radii resulting from deformations related to the strain gradients in released straight test structure. The test structures are fabricated according to the description above. The influence of wet etchants versus dry etchants is illustrated, whereby wet etchants may preferably be  $\text{NH}_4\text{HF}+\text{HC}_2\text{H}_3\text{O}_2+\text{H}_2\text{O}$  or 6-20:1 buffered HF. The table below shows, that the curvatures get smaller essentially as the square of the sputtered layer thickness for otherwise constant remaining first sputtering criteria.

Sputtered Layer Thickness	Curvature of initial test structure ( $\text{mm}^{-1}$ )	Curvature of 350C annealed test structure ( $\text{mm}^{-1}$ )
0.6 $\mu\text{m}$	Wet release: 0.50 +/- 0.09 Dry release: -1.21 +/- 0.06	Wet release: 0.27 +/- 0.13
2.0 $\mu\text{m}$	Wet release: 0.04 +/- 0.01 Dry release: -0.20 +/- 0.06	Wet release: 0.01 +/- 0.02
5.0 $\mu\text{m}$	Wet release: 0.01 +/- 0.03 Dry release: 0.01 +/- 0.01	Wet release: 0.01 +/- 0.02
Wet release used in combination with oxide sacrificial layers and HF-based etchants Dry release used in combination with polyimide sacrificial layers and an Oxygen plasma etchant.		

For curvature radii much greater than the film thickness, the average strain gradient in the released structures is merely the inverse of the curvature radius and is seen to decrease with increasing thickness and low temperature annealing.

The secondary sputtering criteria for sputtered boron doped silicon are preferably a first sputtering power in the range of 1.5-2.5kW, a working gas pressure of 8-14mTorr

argon, as well as the sputtered layer thickness and the selected sacrificial layer material. For these second sputtering criteria, the sputtering rate ranges between 19-37nm/min. The power is spread over a target area of approximately 6 by 10 inches.

Experimentation has shown that the initial in-plane strain is influenced by the sacrificial layer material on which the boron doped silicon is sputtered. The third table below shows the initial strain values for second sputtering criteria including the combination of sputtered silicon deposited on PSG. This table is a function of the stress as a function of deposition settings rather than a function of substrate type.

Power	Stress for 8mTorr	Stress for 14mTorr
1.5 kW	97 MPa Tensile	106 MPa Tensile
2.5 kW	27 MPa Tensile	133 MPa Tensile

The fourth table below shows the initial strain values for second sputtering criteria including the combination of sputtered silicon deposited on bare silicon.

Power	Stress for 8mTorr	Stress for 14mTorr
1.5 kW	34 MPa Tensile	141 MPa Tensile
2.5 kW	22 MPa Compressive	164 MPa Tensile

The fifth table below shows initial and residual strain values for varying boron doped silicon layer thickness sputtered with a sputtering power of 2.0kW and an ambient sputtering pressure of 9.5mTorr. The values of the fourth  
5 table are illustrated for a defined low temperature annealing process at 350°C for 3 hours in forming gas in the preferred configuration of 1:9 H<sub>2</sub>:N<sub>2</sub>.

Sputtered layer thickness	Initial stress value	Residual stress values
0.6 μm	95-108 MPa Tensile	95-117 MPa Tensile
2.0 μm	102-107 MPa Tensile	69-76 MPa Tensile
5.0 μm	110-117 MPa Tensile	74-77 MPa Tensile

To prevent buckling of released micro-machined structures supported in a buckling critical fashion as is known to those skilled in the art, residual in-plane strain needs to be tensile. In case, where the residual in-plane strain  
10 remains compressive, below a buckling threshold.

An initial surface roughness as part of the pre-annealing configuration is in a predetermined relation to a residual  
20 surface roughness as part of the post-annealing configuration. Surface roughness is an important consideration for bearing and optical surfaces. The sixth table below shows the exemplary surface roughness of three wafers of varying sputtered silicon thickness, before and  
25 after low temperature annealing at 350°C:

Sputtered layer thickness	Initial surface roughness	Residual surface roughness
0.6 $\mu\text{m}$	4.2 nm	3.0 nm
2.0 $\mu\text{m}$	4.9 nm	3.1 nm
5.0 $\mu\text{m}$	5.9 nm	4.9 nm

5 An initial electrical resistivity as part of the pre-annealing configuration is in a predetermined relation to a residual electrical resistivity as part of the post-annealing configuration. The seventh table below shows the initial electrical resistivity and the residual electrical resistivity for varying sputtered layer thickness.

10

Sputtered layer thickness	Initial electrical resistivity	Residual electrical resistivity
0.6 $\mu\text{m}$	50 M $\Omega/\square$	125 G $\Omega/\square$
2.0 $\mu\text{m}$	20 M $\Omega/\square$	20 G $\Omega/\square$
5.0 $\mu\text{m}$	7 M $\Omega/\square$	6 G $\Omega/\square$

The table shows, that the residual electrical resistivity is significantly higher than the initial electrical resistivity. For applications where a low residual electrical resistivity is required, one or more additional layers of a highly conductive material may be sputtered on top and/or below the sputtered layer. **Figs. 6-9** illustrate schematically an exemplary fabrication process including

the sputtering of a conductive bottom layer **Ti1** (see **Figs. 8, 9**) and a conductive top layer **Ti2** (see **Figs. 8, 9**).

The conductive bottom and top layers **Ti1**, **Ti2** may be patterned in the same etchant as the resistive sputtered silicon core layer **13** (see **Fig. 8**). The compatible dissolving characteristic of the bottom layers **Ti1**, **Ti2** and the core layer **13** allow a simultaneous etching to shape the micro-machined structure **131** (see **Figs. 8, 9**).

The bottom and/or the top layer **Ti1**, **Ti2** may be sputtered with differing sputtering criteria resulting in predetermined differing residual in-plane strain and/or differing residual strain gradients that differ from the residual core strain and residual core strain gradient. Hence, the post-annealing residual strain gradient and or in-plane strain of the released structure **131** may be adjusted by using differing sputtering criteria.

**Fig. 5** shows a simplified block diagram illustrating the alternating selections of sputtering criteria dependent on the deformation configuration of the micro-machined structures. After designing a micro-machined structure, it is determined whether Euler's buckling principles need to be considered or not. In case they do, for example for a micro-machined structure in the form of a straight beam rigidly supported on both ends, the buckling-influenced deformation configuration is recognized and second sputtering criteria are selected. After releasing the micro-machined structure, it has a residual in-plane strain within a range where no buckling occurs. In the case, where Euler's buckling principles do not apply, for example

for a micro-machined structure in the form of a straight beam rigidly supported only on one end, the first sputtering criteria are selected. After releasing the micro-machined structure, it has a residual strain gradient within a range where residual deformation remains within a predetermined limit.

*Fig. 6* shows a simplified section between the breaking lines **B3** and **B4** of a work piece according to an embodiment of the present invention of multi layered micro-machined structures. The work piece has aluminum-metalized circuitry elements **A1** on top of an oxidized layer **Ox** covered with a sacrificial layer **12** with an aluminum hard mask **15** on top of it. The sacrificial layer **12** is preferably a polyimide resin spun on, cured and etched back to  $2.0\mu\text{m}$ . The curing is performed at  $350^{\circ}\text{C}$  for 1 hr. The thermal curing budget of the curing process is below the critical thermal budget.

Polyimide may be patterned via an aluminum hardmask **15** or by other techniques as are known to those skilled in the art. The aluminum hard mask **15** may be of 200nm thickness. Due to the back etching of the sacrificial layer **12**, the aluminum hard mask **15** may have an improved adhesion to the sacrificial layer **12**. The aluminum hard mask **15** is patterned for a following primary etching **EC1** (see **Fig. 7**) of contact vias. To reduce the risk of overetching or undercut etching, a photoresist used to pattern the aluminum hard mask **15** (not shown) is not removed prior to the first etch in order to provide a more uniform loading during the etch. In addition, the anisotropy of the oxygen

plasma etch may be increased by reducing the pressure to a practical limit of a commonly used plasma etcher. This level may be, for example, 100mTorr.

- 5 **Fig. 7** shows a simplified section of the work piece of **Fig. 5** after a primary etching. The aluminum hard mask **15** is removed by an aluminum wet etch.

10 In a following step, the conductive bottom layer **Ti1** in the form of low stress TiW with  $\sigma < 100$  MPa may be sputtered in a thickness of 50nm. On top of the conductive bottom layer **Ti1** is the sputtered core layer **13**, in this example sputtered from silicon with a thickness of 2.0 $\mu$ m. On top of the sputtered layer **13** is the conductive top layer **Ti2**, in  
15 this example sputtered also from the same material as the conductive bottom layer **Ti1** with the same thickness of 50nm.

20 Due to the symmetric configuration of the multi layer including the conductive bottom and top layers **Ti1**, **Ti2** and the core layer **13** between, bimorph effects are kept to a minimum.

25 After applying a defined low temperature annealing process with a maximum temperature of 300°C for about 1hr, the multi layer may be exposed to a secondary etching process EC2 (see **Fig. 8**), which may be performed by the use of SF<sub>6</sub>-C<sub>2</sub>ClF<sub>5</sub> plasma. During the secondary etching process **EC2**, the multi layer structure **131** is defined.

30



The electrical resistivity of the multi layer structure 131 may be about 25  $\Omega/\square$ .

In a tertiary etching process, EC3 (see Fig. 9) the  
5 sacrificial layer 12 is dissolved.

With the fabrication method described in the Figs. 6-9, various integrated structures including mechanical elements and electronic circuitry may be fabricated. For example,  
10 an electrostatically actuatable test structure combined with a CMOS capacitance test circuitry may be fabricated. The threshold shifts in nmos and pmos transistors of the pre-fabricated CMOS remain essentially unaltered during the consecutive fabrication of the micro-machined structures.

15 The fabrication of sputtered silicon structures is accomplished with a thermal fabrication budget including a thermal sputtering budget and a thermal annealing budget for the optional low temperature annealing process. The  
20 thermal fabrication budget is lower than a first critical thermal budget of eventual circuitry and/or a second critical thermal budget of the sacrificial layers 2, 12. As a result, sacrificial layers 2, 12 may be made from organic material such as polyimide that are dissolvable  
25 with a dry etch such as oxygen plasma and stiction of the released structure due to the capillary forces of a drying wet etchant is avoided.

Accordingly, the scope of the invention described in the  
30 specification above is set forth by the following claims and their legal equivalent: